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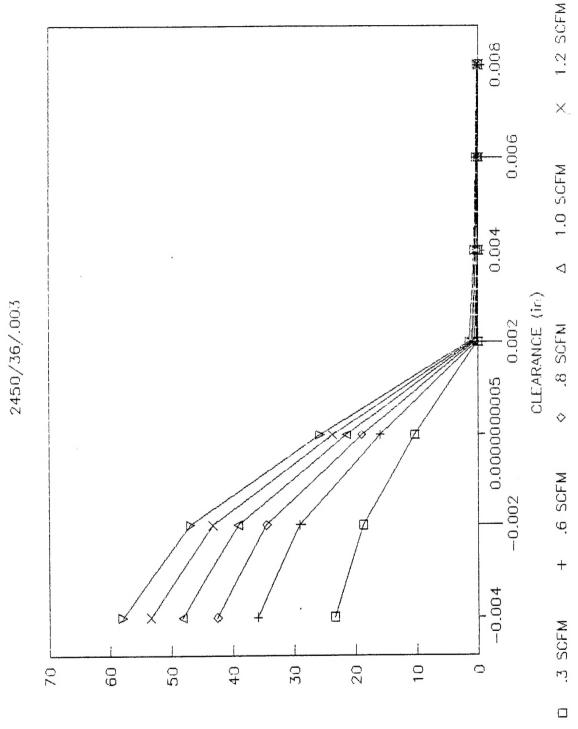
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INTRODUCTION AND BACKGROUND

Brush seals are currently being used for replacement of labyrinth type shaft seals, since brush seals provide better leakage characteristics. Brush seals also allow for shaft dynamics or shaft excursions. When a labyrinth seal experiences a shaft excursion, a permanent groove or scar is left in the seal. presence of a scar on the labyrinth will allow additional leakage, which with time will continue to degrade the seal's effectiveness. A brush seal will bend with the excursion and then return to its original configuration and leakage characteristics. The bristles ride or rest on the shaft of a rotating member, creating the sealing or flow restriction that makes the brush seal a better seal. As the shaft rotates, a tribological problem arises between the bristles and the shaft. Frictional heating of the bristle causes, or at least increases, the oxidation of the bristle alloy and increases wear of the fiber. The net effect of this is to decrease the brush seals' effectiveness. When a brush seal is initially installed; an interference condition is intentionally The clearance between the shaft and the bristles is typically -.010 inches, but after only a few hours of operation the bristles have worn to a zero clearance fit. Bristle wear is caused by frictional heating and related oxidation, which then chips off. In an attempt to alleviate wear, ceramic bristle brush seals are being investigated. Ceramics are well known for their favorable temperature and wear characteristics. A ceramic fiber will allow the bristle to maintain the interference fit for a longer time. With the fibers intimate contact with the shaft, greater brush seal performance can be taken advantage of, for a much longer time. is known that the greater the contact pressure of the bristles on the shaft, the greater the leakage resistance of the brush seal. This tendency of brush seals has been documented by several sources. For this reason, it is desirable that the bristles maintain their interference fit with the shaft over the life of the seal.

Since it has been shown that a brush seal can be made with ceramic fibers (thereby producing a prolonged intimate contact between bristle and shaft), the next logical system member for improvement is the shaft. Most current applications and test programs utilize some type of shaft coating for the bristle to slide against. There are two main reasons for this. The first is to eliminate damage that the brush may do to a bare shaft. safety reasons, it is better that the seal wear, rather than the The second, is to provide a smoother surface for the bristle to ride on, so as to prolong the seal's life. To reduce wear, the coatings are usually required to have a 10μ in. Ra or The polished finish reduces the friction better finish. coefficient, thereby reducing wear. While this does prolong the intimate contact, the usual time required to produce a line-to-line or zero clearance fit is still in the tens of hours. It is acknowledged that even at line-to-line contact a brush seal still reduces leakage substantially better than a labyrinth but with an interference fit, a brush seal can even do better. Figure 1 is the

TECHNETICS TEST RIG



Typical Graph of Predicted Brush Seal Performance

Figure 1

1.4 SCFM

PRESSURE DROP (bars)

predicted leakage of a typical brush seal on a circumferential inch basis. It is seen that the seal's performance drops off sharply with a clearance increase between the seal's bristles and the shaft - evidencing the performance advantages that could be realized if a bristle and coating can be developed to retain an interference fit.

TECHNICAL OBJECTIVES

The technical objectives of this work were:

- 1. Identify candidate coatings
- 2. Identify and/or develop coating application methods
- 3. Screen coatings and application methods

In the identification of candidate coatings, coefficients of friction, wear, temperature and capabilities/properties were considered - to find the best available state of the art coatings which may compliment the brush seal system. Coatings that were examined are industry standard coatings and variants thereof.

Experimental work indicates that the wear of a ceramic can be attributed, in part, to its porosity. More porosity produces more wear. For this reason coating application methods were sought, which would produce a low porosity structure. It has also been demonstrated that ceramics with a softer second phase around the grain boundaries may also reduce wear. A coating, produced by cospraying two materials to produce a soft second phase, was evaluated. It was believed that a co-sprayed coating of zirconia and boron nitride could produce a softer second phase surrounding hard primary phase grains, thus producing a dry lubricated coating with good potential.

Screening of the coatings and application methods was based on properties and characteristics known to be necessary for a low wear tribological system. Considered candidate coatings must have had at least some favorable properties that fit the goals of this effort. For this reason, a literature search, as well as discussions with industry leaders in coatings, was completed prior to choosing coatings for testing. Screening proceeded with temperature, oxidation, thermal shock, wear and friction testing.

WEAR TESTING

Wear testing consisted of placing a prepared tuft of fiber in contact with a high speed rotating rotor on which a coating had been applied. Figure 2 shows a simplified view of the test apparatus. The rig is powered by a 15 hp air turbine capable of speeds up to 50,000 rpm. For purposes of these tests, a heavier 6 inch diameter rotor was employed and rotational speeds were held to a maximum of 40,000 rpm. The tuft holder is attached to an air bearing to minimize any frictional error. A moment arm is

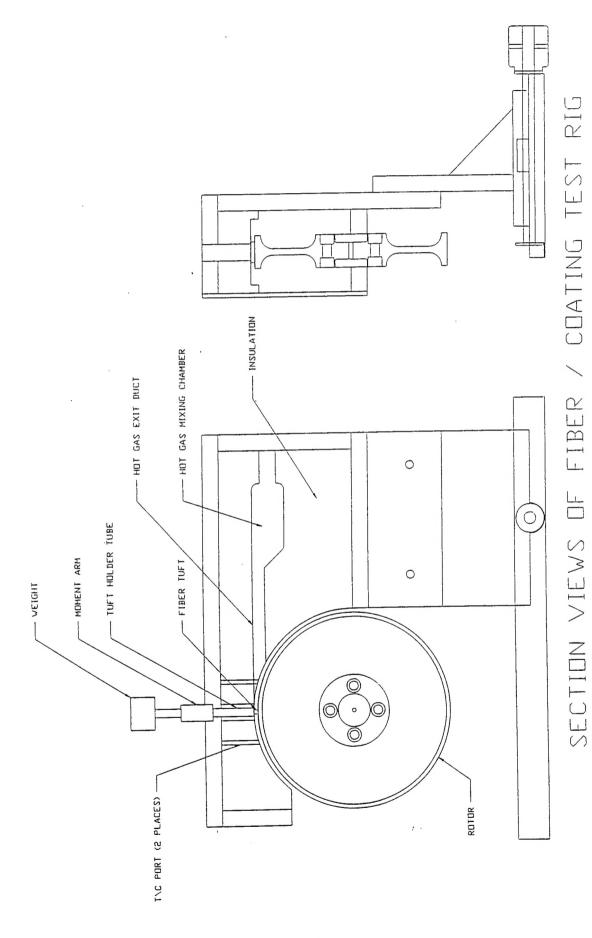


Figure 2

projected off the air bearing shaft and rests on a one pound load cell. By adjusting the load cells position in relation to the axis of the air bearing, the load sensitivity can be adjusted. Additional test points were temperatures taken by thermocouples placed for and aft of the tuft/rotor contact point.

Prior to testing, a rotor was prepared by applying and polishing a coating. A super finishing process was used for polishing. Surface finishes were 10 Ra or below, when possible. After polishing, rotors were mounted on the air turbine and dynamically balanced. Rotors were balanced to below 0.00002 inches peak to peak vibration. Tufts were prepared by placing approximately 350 SiC fibers of 0.006 inch diameter into a ceramic tube and fixing the fibers with ceramic cement. At least 0.75 inches of fiber was left protruding from the tube and was used for, or consumed in testing. Similar tuft samples were also made from Haynes 25 fiber.

Testing began by bringing the rotor up to speed and letting it stabilize. Base readings for temperature and drag load were then taken. A fiber tuft was applied with a dead load, normal to a tangent line from the rotor with the tuft axis directly over the center of the rotor. Three sets of readings were taken. Initial readings were taken right after application of the tuft. The second reading was taken at 3 minutes. A final set of readings was then taken at the conclusion of the test after a total of six minutes of fiber to coating/rotor contact. Tuft wear measurements were taken prior to increasing the dead load. Combination of 3 weights and 3 speeds were made to develop the test matrix. Figure 3 shows the test matrix.

THERMAL SHOCK TESTING

A coupon with coating material applied was thermally shocked for 2,178 cycles to determine if thermal cycling would cause the coatings to crack. A coupon consists of a superalloy sheet metal rectangle one inch by three inches. It is rolled to a convex contour. The coupon is then fixtured next to the rotor during coating application. Testing of these coupons consisted of simultaneously heating the coating surface to 2000°F while cooling the back side with a 150 SCFH air jet. The coupon is then quenched with a 150 SCFH air jet on the coating side. Each cycle lasts 90 seconds. Periodic examinations are made for sampled degradation.

COATINGS

Six coatings were chosen for testing. Chosen coatings were all thought to have desirable characteristics for a ceramic brush seal application.

1. <u>Chrome Carbide</u> - This coating was supplied by General Plasma and is their coating GPX-2176HP. Composition is 75% chromium carbide and 25% 80 nickel/20 chromium. It is applied by plasma flame spray. This and similar

Test Matrix

TUFT TESTING

REPEATABILITY STUDY

03/28/95

BN/PSZ COATED ROTOR

| | | S20 | M20 | L20 | S30 | M30 | L30 | S40 | M40 | L40 |
|-------------------|--------------|--|---------|----------|----------------------|-----------------------|--|---------------------------|---------|---------|
| | | TRAC/PN | TRAC/PN | TRAC/PN | | TRAC/PN | TRAC/PN | TRAC/PN | TRAC/PN | TRAC/PN |
| SiC Fibers | AMB | 9A/038 | 9A/038 | 9A/038 | 9A/038 | 9A/039 | 9A/039 | 9A/040 | 9A/040 | DISC |
| SiC Fibers | AMB | 9A/038 | 9A/038 | 9A/038 | 9A/039 | 9A/039 | 9A/039 | 9A/040 | 9A/040 | DISC |
| SiC Fibers | AMB | 9A/038 | 9A/038 | 9A/038 | 9A/039 | 9A/039 | 9A/039 | 9A/040 | 9A/040 | DISC |
| SiC Fibers | AMB | 9A/038 | 9A/038 | 9A/038 | 9A/039 | 9A/039 | DISC | 9A/040 | 9A/040 | DISC |
| SiC Fibers | COLD | 9B/041 | 9B/041 | 9B/041 | 9B/042 | 9B/042 | 9B/042 | 9B/043 | 9B/043 | 9B/044 |
| SiC Fibers | COLD | 9B/041 | 9B/041 | 9B/041 | 9B/042 | 9B/042 | 9B/042 | 9B/043 | 9B/043 | 9B/044 |
| SiC Fibers | COLD | 9B/041 | 9B/041 | 9B/042 | 9B/042 | 9B/042 | 9B/043 | 9B/043 | 9B/043 | 9B/044 |
| SiC Fibers | COLD | 9B/041 | 9B/041 | 9B/042 | 9B/042 | 9B/042 | 9B/043 | 9B/043 | 9B/043 | 9B/044 |
| | | | (| PSZ COAT | ED ROTOR | h | | | | |
| 0:0 EI | | | 0.000 | 04 1000 | 8B/037 | 8B/037 | 8B/037 | 8C/037 | 8C/037 | DISC |
| SiC Fibers | AMB | 8A/036 | 8A/036 | 8A/036 | | 8B/037 | 8B/037 | 8C/037 | 8C/037 | DISC |
| SiC Fibers | AMB | 8A/036 | 8A/036 | 8A/036 | 8B/037 | 8B/037 | 8B/037 | 8C/037 | 8C/037 | DISC |
| SiC Fibers | AMB | 8A/036 | 8A/036 | 8A/036 | 8B/037 | Market and the second | 8B/037 | 8C/037 | 8C/037 | DISC |
| SiC Fibers | AMB | 8A/036 | 8A/036 | 8A/036 | 8B/037 | 8B/037 | 05/03/ | 6C/001 | 90/001 | Dioc |
| | | | | CHROME (| CARBIDE C | OATED RO | TOR | | | |
| SiC Fibers | AMB. | 7A/019 | 7A/019 | 7A/019 | 7A/019 | 7A/019 | 7A/019 | 7A/020 | 7A/020 | 7A/020 |
| SiC Fibers | AMB | 7A/019 | 7A/019 | 7A/019 | 7A/019 | 7A/019 | 7A/019 | 7A/020 | 7A/020 | 7A/020 |
| SiC Fibers | AMB | 7A/019 | 7A/019 | 7A/019 | 7A/019 | 7A/019 | 7A/019 | 7A/020 | 7A/020 | 7A/020 |
| SiC Fibers | AMB | 7A/019 | 7A/019 | 7A/019 | 7A/019 | 7A/019 | 7A/019 | 7A/020 | 7A/020 | 7A/020 |
| SiC Fibers | COLD | 7B/025 | 7B/025 | 7B/025 | 7B/025 | 7B/025 | 7B/025 | 7B/025 | 7B/025 | 7B/025 |
| SiC Fibers | COLD | 7B/025 | 7B/025 | 7B/025 | 7B/025 | 7B/025 | 7B/025 | 7B/025 | 7B/025 | 7B/035 |
| SiC Fibers | COLD | 7B/025 | 7B/025 | 7B/025 | 7B/025 | 7B/025 | 7B/025 | 7B/025 | 7B/025 | 7B/035 |
| SiC Fibers | COLD | 7B/025 | 7B/025 | 7B/025 | 7B/025 | 7B/025 | 7B/025 | 7B/025 | 7B/025 | 78/035 |
| | | | | BARE ROT | OR – NO C | COATING | | | | |
| SiC Fibers | AMB | 10A/050 | 10A/050 | 10A/050 | 10A/050 | 10A/050 | 10A/050 | 10A/050 | 10A/050 | 10A/051 |
| SiC Fibers | AMB | 10A/050 | 10A/050 | 10A/050 | 10A/050 | 10A/050 | 10A/050 | 10A/050 | 10A/051 | 10A/051 |
| SiC Fibers | AMB | 10A/050 | 10A/050 | 10A/050 | 10A/050 | 10A/050 | 10A/050 | 10A/050 | 10A/051 | 10A/051 |
| SiC Fibers | AMB | | 10A/050 | 10A/050 | 10A/050 | 10A/050 | 10A/050 | 10A/050 | 10A/051 | 10A/051 |
| SIC Fibers | AMID | THE TOTAL OF THE PARTY OF THE P | 10/4000 | | airii kar Yaraa aala | | imilia e e e e e e e e e e e e e e e e e e e | h Made our de casar au de | | |
| | | | | BARE ROT | OR – NO C | COATING | | | | |
| H25 Fibers | AMB | 10B/052 | 10B/052 | 10B/052 | 10B/052 | 10B/052 | 10B/052 | 10B/052 | 10B/052 | 10B/053 |
| H25 Fibers | AMB | 10B/052 | 10B/052 | 10B/052 | 10B/052 | 10B/052 | 10B/052 | 108/052 | 10B/052 | 10B/053 |
| H25 Fibers | AMB | 10B/052 | 10B/052 | 10B/052 | 108/052 | 10B/052 | 10B/052 | 10B/052 | 10B/052 | 10B/053 |
| H25 Fibers | A M B | 10B/052 | 10B/052 | 108/052 | 10B/052 | 10B/052 | 10B/052 | 10B/052 | 10B/052 | 10B/053 |

| WEIGHTS | RPM'S | COMPLETED | DISC - DISCONTINUED |
|------------|----------------------------|--|-----------------------------|
| S - SMALL | 20 - 20,000 | AMBIENT - NO AIR FLOW COLD - 7.5 SCFM | SIC - SILICON CARBIDE FIBER |
| M - MEDIUM | 30 - 30,000 40 - 40,000 | COLD = 7.3 SCFM | H25 - HAYNES 25 FIBER |

EXAMPLE: 1A/001 - ROTOR:1; TRACK:A; TUFT:001

coatings are the current coatings of choice for metallic brush seal applications. The coating proved very hard and dense. These attributes enabled the coating to be polished to a finish of less than one micro inch Ra.

- Boron Nitride filled PSZ A plasma co-sprayed mixture of boron nitride and yttria stabilized zirconia was developed for this program in an effort to provide an oxide coating with a dry high temperature lubricant additive. It was known that a two phase system composed of a hard primary phase and a soft secondary phase would produce a coating resistant to crack propagation. It is also known that born nitride can provide a high temperature lubricant. These two characteristics are believed to be necessary for a successful ceramic brush seal coating.
- 3. Partially Stabilized Zirconia A plasma sprayed yttria stabilized zirconia was chosen for its fracture toughness and the non-stick or slick properties of zirconia. Zirconia has a high coefficient of thermal expansion so it lends itself well to the direct application onto metals. Ytterbia stabilized ceramics are known for their fracture toughness. These characteristics would eliminate the cracking and high wear situation seen in alumina coatings.
- 4. Partially Stabilized Zirconia A vapor deposited PSZ was chosen for its high density and previously noted properties. This coating was the highest density of all coatings chosen. This coating was not available at the time of testing, due to coating application problems. The coating was incompatible with the high expansion rotor material. Bond coat failures on two attempts precluded any further testing.
- 5. <u>Alumina</u> A HVOF applied alumina coating was chosen for its high density and oxidation resistance. This coating was not available at the time of testing.
- 6. <u>Triboglide®</u> This coating is known for its high temperature, self-lubricating properties. It has been tested extensively for bearing applications and initial testing for application with metallic brush seals has shown excellent results. This coating was not available at the time of testing.

RESULTS

Raw wear data with calculated values and plotted data are provided in Appendix A. Thermal shock testing was done on the oxide ceramic coatings to determine if thermal growth or shock would cause cracking or spallation. All passed 2,178 cycles

without one crack or spallation. Ultimate tensile strength testing of the oxide ceramic coatings had similar results in that all coatings exceeded the minimum tensile strengths for test conditions.

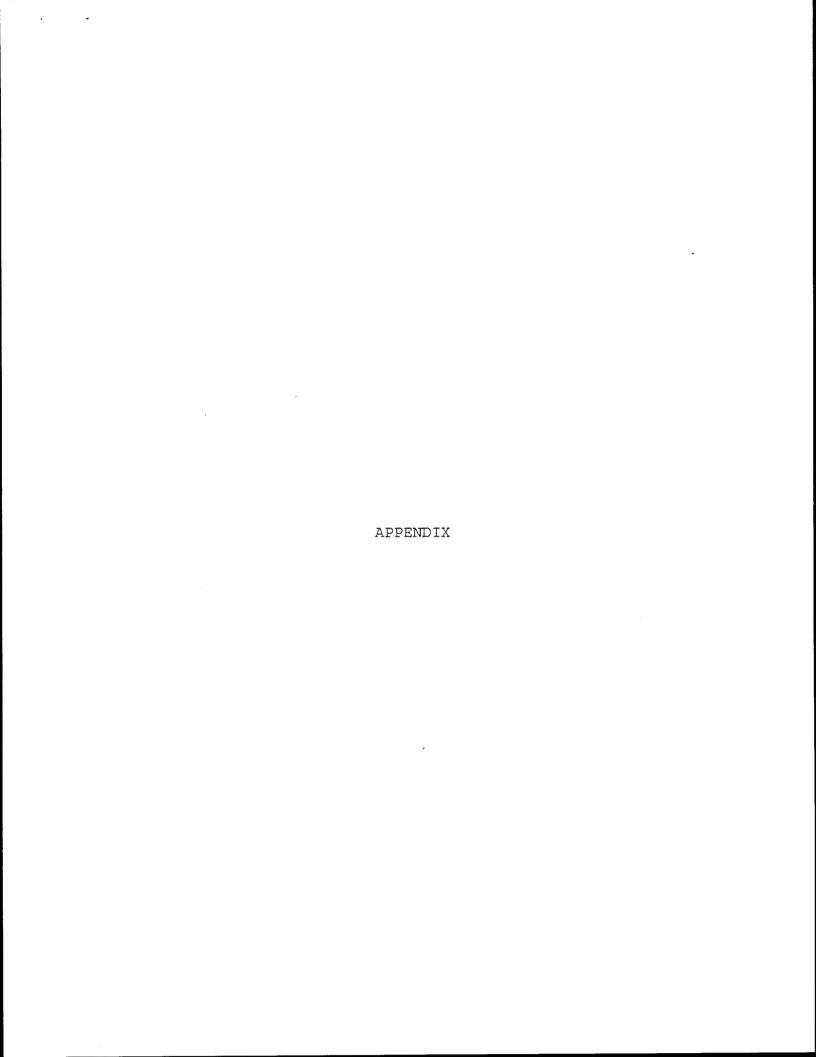
Wear testing was the primary objective of this program and generated the most applicable data for ceramic brush seals. The chrome carbide coating suffered the most damage. After only a few minutes of testing, the SiC fiber tufts wore through the .006-.007" thick coating. It is believed that the metal matrix oxidized away and the carbide particles were then free to fall off. Metal matrixes or bonding agents cannot withstand the high temperatures created by friction between the ceramic bristles and the coating, even after the coating has been polished to a mirror finish.

The oxide ceramic coatings did not fair much better than the chrome carbide. Life was about tripled, but deep scars appeared after only a relatively short time. The condition of dry unlubricated sliding friction between two hard ceramics produced high wear.

Bare or uncoated rotors made of 17-4PH proved to be the most wear tolerant material tested. It is believed that the 17-4PH smeared rather than fractured off, like the harder more brittle ceramics. Wear caused by oxidation of the metal was probably reduced because of the metals' greater thermal conductivity, absorbing and conducting away the frictional heat. Low total test time also contributed to the lack of oxidation wear.

CONCLUSION

No coating tested would be suitable for a rotating ceramic brush seal application. Wear was too high in all cases. However, bare uncoated metallic surfaces with 100% densities could prove to be good applications for low speed and low temperature ceramic brush seals. Investigation of coatings containing lubricants must be tested to see if they can provide the needed low wear additive required for an increase in brush seal performance.



| PART #9 TRAC A | V | | | | | | BN/PSZ | BN/PSZ COATED ROTOR | ROTOR | | | | | | |
|--|-------------|------------------|--------|-------|--------|----------|-----------|---------------------|-------|-----------------------|----------|--------|--------|--------|------------|
| TUFT 039 START THRU L30(1ST) TUFT 039 L30(2ND) THRU L30(4TH) | IT THRU L30 | (1ST) 30(4TH) | | | | | AMBIENT | ambient air testing | r. | | | | | | |
| TUFT 040 S40 THRU THE END FIBER/ END | HRU THE EI | STARI | TEMP | | START | TEMP | END | ST | DIE | START | END | TW SO | DRAG | Z | Ö |
| IES! | 102(1) | (CZ(T) | CHANGE | - | 103177 | CHANGELT | 0 | 2000 | il_ | CLIMINGE IF I WILLIAM | 19 4 572 | 0.0116 | 0.4250 | 9013 | 1 0984 *** |
| SIC/SZU/AMB | 10.2 | 05.3 | 000 | 200 | 20.00 | 2.5 | 1 | | | | 13 1554 | 0.0018 | 0.4522 | 0.3913 | |
| | 126.7 | 1000 | 24.7 | | 87.8 | 24. | | | 0.259 | 1 | 13.1516 | 0.0038 | 0.5253 | 0.3913 | 1.3424 *** |
| | 129.6 | 101.2 | 28.4 | 110.6 | 85.2 | 25.4 | | 0.005 | 0.241 | | 13.1489 | 0.0027 | 0.4887 | 0.3913 | 1.2491 *** |
| SiC/M20/AMB | 1297 | 104.2 | 25.5 | | 86.2 | 21.8 | 8 0.241 | 0.007 | 0.234 | 13.1489 | 13.1436 | 0.0053 | 0.4746 | 0.6017 | 0.7887 |
| OLO MICOLOMIC | 126.0 | 107.3 | 187 | | 86.7 | 19.8 | 1 | | | | Ī | 0.0043 | 0.4725 | 0.6017 | 0.7853 |
| | 128.0 | 101.7 | 26.3 | | 94.1 | 23 | | | | 13.1393 | ľ | 0.0035 | 0.4908 | 0.6017 | 0.8157 |
| | 128.5 | | 23.1 | 109.2 | 86.4 | 22.8 | 0.266 | | | | | 0.0069 | 0.5273 | 0.6017 | 0.8764 |
| SiC/L20/AMB | 142.2 | 71.7 | 70.5 | | 76.9 | | | | | | 13.0977 | 0.0312 | 0.7341 | 1.1172 | 0.6571 |
| | 127.2 | 110.3 | | | 9.98 | | | | | . ! | | 0.0139 | 0.5739 | 1.1171 | 0.5137 |
| | 122.8 | 103.6 | 19.2 | 112.8 | 86.0 | 26.8 | | 600.0 | 0.275 | 13.0838 | 13.0660 | 0.0178 | 0.5577 | 1.1171 | 0.4992 |
| | 122.9 | 104.2 | ! | | 99.9 | | 0.307 | | j | j | | 0.0216 | 0.6104 | 1.1171 | 0.5465 |
| SiC/S30/AMB | 112.7 | 102.7 | 10.0 | 1 | 96.0 | | | | | | | 0.0044 | 0.1744 | | 0.4460 |
| | 111.1 | | 9.2 | 104.9 | 96.8 | | | | 0.083 | 13.0400 | ΤÏ | 0.0022 | 0.1683 | 0.3910 | 0.4305 |
| | 110.2 | 102.4 | | | 96.9 | 7.7 | | | | | _ | 0.0016 | 0.1622 | Ì | 0.4149 |
| | 112.3 | | | | 96.8 | | 0.096 | 0.005 | 0.091 | 13.0362 | 13.0340 | 0.0022 | 0.1845 | 0.3910 | 0.4720 |
| SiC/M30/AMB | 118.7 | 101.8 | | | 96.9 | 14.6 | 6 0.135 | 0.007 | 0.128 | 13.0340 | 13.0281 | 0.0059 | 0.2596 | 0.6014 | 0.4316 |
| | 119.4 | 106.9 | | | 98.9 | 12. | | 0.006 | | 13.0281 | 13.0205 | 0.0076 | 0.2515 | 0.6014 | 0.4181 |
| | 117.3 | | 10.9 | | 98.0 | 12.0 | 0 0.123 | | 0.120 | | 13.0130 | 0.0075 | 0.2434 | 0.6014 | 0.4047 |
| | 116.8 | | 7 | 109.8 | 97.4 | 12. | 4 0.118 | 90000 | | 13.0130 | 13.0060 | 0.0070 | 0.2292 | 0.6014 | 0.3811 |
| SiC/L 30/AMB | 129.5 | | | 124. | 97.0 | 27.0 | 0 0.226 | 0.004 | 0.222 | | 12.9580 | 0.0480 | 0.4502 | 1.1169 | 0.4031 |
| | 123.5 | 67.8 | 55.7 | | 83.9 | | | | 0.251 | 12.9372 | 12.8987 | 0.0385 | 0.5090 | 1.1167 | 0.4558 |
| Machiner restreet sections of the states of the section of the sec | 121.6 |) old | Join | 118.1 | 88.5 | 29 | 0.239 | 0.006 | 0.233 | 10 | 12.8432 | 0.0555 | 0.4725 | 1.1166 | 0.4232 |
| | 200 | 200 | 200 | 2000 | 2 | | 22 | 200 | 2000 | 200 | | | | | |
| SiC/S40/AMB | 114.9 | | | | | | | | | | | 0.0291 | 0.1744 | 0.3925 | 0.4444 |
| | 117.0 | 105.5 | 11.5 | 107.9 | 101.5 | | 6.4 0.114 | 0.006 | 0.108 | 13.6955 | 13.6891 | 0.0064 | 0.2190 | 0.3925 | 0.5581 |
| | 117.2 | | | | | | | | | | | 0.005 | 0.1300 | 0.3823 | 0.4057 |
| | 124.1 | | 17.8 | | \$ | 0 | | | 0.100 | 13.0040 | 13.0/01 | 60000 | 0.2123 | 0.3324 | 0.5420 |
| SiC/M40/AMB | 123.7 | 107.0 | 16.7 | 113. | 102.2 | | | | | | | 0.0127 | | | 0.5013 |
| | 121.2 | | 11.0 | | 101.6 | 6.6 | 9 0.140 | 0.006 | 0.134 | 1 | | 0.0156 | 0.2718 | | 0.4508 |
| | 123.1 | | | 115. | 104.0 | | | | | 13.6498 | | 0.0142 | | | 0.5114 |
| | 121.1 | 109.8 | | 114 | 105.0 | 6 | _ | | | | 13.6240 | 0.0116 | | 0.6027 | 0.4206 |
| SiC/L40/AMB | DISC | DISC | DISC | DISC | DISC | DISC | DISC | DISC | DISC | DISC | DISC | DISC | DISC | DISC | DISC |
| | DISC | DISC | DISC | DISC | DISC | DISC | DISC | DISC | DISC | DISC | DISC | DISC | DISC | DISC | DISC |
| | DISC | DISC | DISC | DISC | DISC | DISC | DISC | DISC | DISC | DISC | DISC | DISC | DISC | DISC | DISC |
| | DISC | DISC | DISC | חמר | Disc | Disc | מפום | אפוח | הומכ | Disc | חפר | Ulac | DISC | DISC | USC |

NOTES: TC2 — Thermocouple located after the tuft slightly above the rotor. TC3 — Thermocouple located before the tuft slightly above the rotor DL — Drag Load (lbs. @ 6.085 in. radius)

SiC - Silicon Carbide Fibers

20 - 20,000 RPM's 30 - 30,000 RPM's 40 - 40,000 RPM's

FILE: TTRPBASI

AMB -- Ambient Air (no flow)

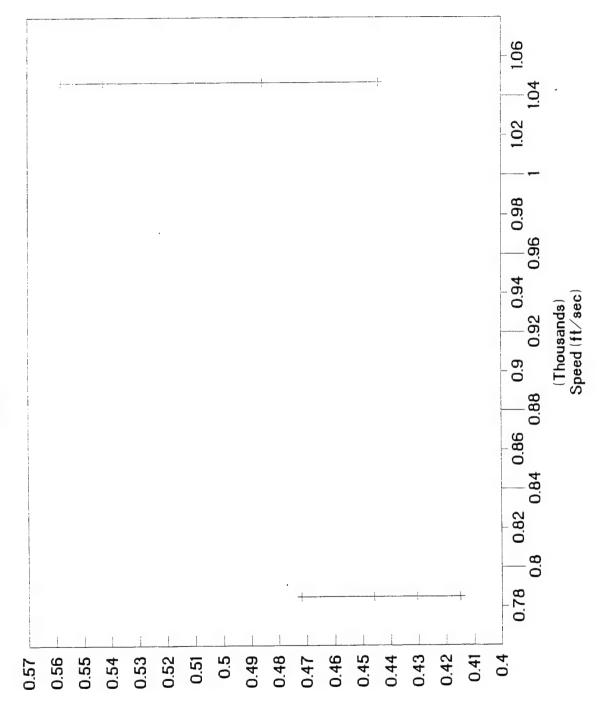
*** - Test Failure

S – Small Weight 164.30g M – Medium Weight 259.73g L – Large Weight 493.56

GF - Coefficient of Friction

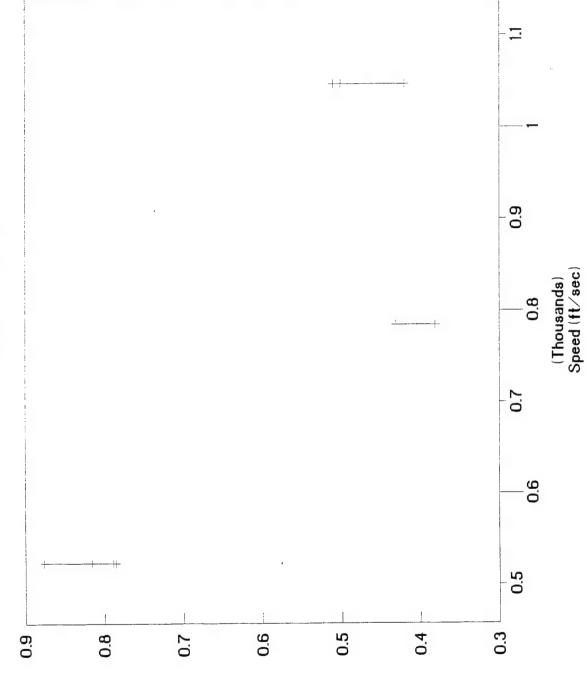
SiC on BN/PSZ

Ambient Air / Small Wt.



CotF

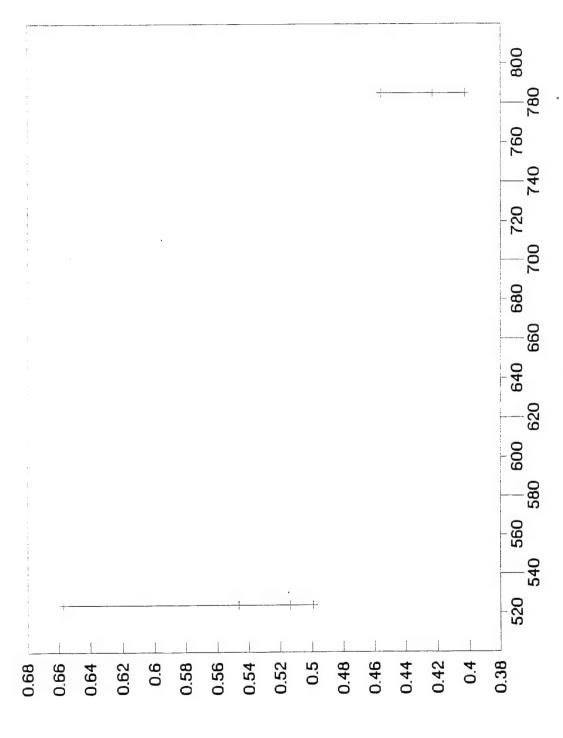
SiC on BN/PSZ Ambient Air / Medium Wt.



CofF

SiC on BN/PSZ

Ambient Air / Large Wt.



Speed (ft/sec)

CofF

BN/PSZ COATED ROTOR

PART #9 TRAC B

TUFT PART #041 START THRU L20(2ND)
TUFT PART #042 THRU L30(2ND)
TUFT PART #043 THRU M40
TUFT PART #044 REMAINDER FIBER/

SiC/M20/CLD

COLD AIR TESTING

13.0718 CHANGE 1FT WT(g) II 0.169 13.143 0.169 13.0755 0.169 13.0755 0.169 13.0754 START 0007 0.009 0.0007 0.010 END DL 0.176 0.162 0.162 CHANGE (°F)
0.3
0.3
0.5
1
-1.7 TEMP START TC3(T) (64.7 64.8 55.1 52.9

0.8761 0.8711 0.8037 0.8763

₽

0.5866 0.5293 0.5125 0.4889

(b) 0.3912 0.3911 0.3911 0.3913 0.3913 0.3913 0.6017 0.6017 0.6016 0.6016 1.1175 0.6015 0.6015 0.6015 0.6015 1.117 1.1170 1.1171 7.11.1 0.3917 0.3917 0.3917 0.3917 0.6021 0.6021 0.6021 0.6020 0.5070 0.3711 0.4157 0.3143 0.2413 0.2515 0.2819 0.3285 0.4522 0.4847 0.4969 0.4969 (lb) 0.3427 0.3143 0.3127 0.3529 0.3184 0.3083 0.2941 0.4887 0.5313 0.5658 0.5658 0.2332 0.1947 0.1886 0.1987 0.2494 0.2434 0.2859 0.2657 0.1724 0.1785 0.1845 0.1825 WT 1.OSS(g) 0.0018 0.0025 0.0021 0.0209 0.0235 0.0156 0.0172 0.0004 0.0013 0.0008 0.0030 0.0062 0.0049 0.0061 0.0125 0.0115 0.0131 0.0194 0.0369 0.0680 0.0386 0.0385 0.0351 0.0396 0.0344 0.0440 0.0068 0.0051 0.0041 0.0026 0.0032 0.0040 0.0019 0.0029 13.3200 13.3151 13.3102 13.1643 13.1592 13.1551 13.1525 13.1285 13.1285 13.0960 13.2513 13.2127 13.1742 13.0389 13.3489 13.3317 13.2590 13.2294 13.2151 13.1711 13.0686 13.0646 13.0627 13.0598 13.3313 13.3292 13.3262 13.0686 13.0686 13.0646 13.3562 13.3193 13.2513 0.096 13.3317 0.096 13.3313 0.093 13.3300 0.096 13.3292 13.3262 13.3200 13.3151 13.3102 13.2690 13.2495 13.2151 13.1711 13.1643 13.1592 13.1551 13.1525 13.1400 13.1285 13.1154 13.0598 13.0389 13.3645 13.3489 0.223 0.239 0.245 0.245 0.085 0.088 0.091 0.119 0.124 0.139 0.162 0.250 0.183 0.205 0.155 0.174 0.157 0.152 0.145 0.241 0.262 0.279 0.279 0.123 0.120 0.141 0.131 0.006 0.010 0.007 0.007 0.007 0.00 0.00 0.00 0.00 0.00 0.009 0.007 0.006 0.008 0.008 0.248 0.269 0.287 0.289 0.123 0.129 0.126 0.147 0.139 0.229 0.252 0.252 0.091 0.095 0.098 0.097 0.128 0.133 0.148 0.169 0.259 0.192 0.213 0.165 0.182 0.165 0.159 0.154 4 0 0 4 -0.3 -0.3 -0.4 0.5 0.4 0.9 0.5 0.5 0.9 0.8 1.6 0.7 1.6 0.5 0.4 0.4 51.5 52.1 52.6 52.4 52.8 53.0 52.8 53.9 54.8 54.8 54.8 55.6 57.0 57.5 57.3 57.5 56.8 58.2 58.4 51.7 50.5 50.0 49.6 48.9 48.9 48.4 49.6 50.7 51.2 51.3 END 1C3(°F) 65.0 65.3 65.3 53.4 51.8 54.7 55.2 55.0 57.0 57.4 57.6 57.7 58.2 58.4 58.7 51.4 51.5 51.8 52.2 52.6 52.8 52.8 53.7 53.8 54.4 54.6 51.4 50.2 49.7 49.2 49.4 49.3 49.6 12.9 7.8 5.4 1.7 33.5 10.8 7.8 8.9 25.1 41.4 11.2 6.3 7.9 7.9 16.2 11.5 16.6 9.6 14.5 13.8 12.1 12.2 19.2 16.1 27.1 15.6 TEMP 11.1 10.2 10.1 6.6 5.5 7.6 7.1 CHANGE 70.4 71.7 72.9 73.5 73.8 76.8 71.9 81.0 79.4 83.6 82.4 79.3 64.3 87.6 87.4 85.3 76.4 63.6 90.4 89.6 63.9 62.0 62.9 61.2 START IC2(°F) (66.5 68.4 59.3 59.3 64.0 64.0 58.8 65.8 63.3 56.3 69.1 65.0 101.5 105.0 101.6 95.9 92.3 91.4 87.8 81.0 97.8 98.4 95.2 94.2 72.0 71.9 75.0 84.2 83.8 85.1 85.6 93.0 92.9 99.0 96.6 END 77.6 78.6 78.6 69.4 69.4 69.6 68.0 68.3 79.9 78.8 78.7 79.5 TEST SiC/S20/CLD SiC/M40/CLD

0.4375 0.4757 0.5062 0.5062

0.5954 0.4971 0.4815 0.5074

0.4143 0.4042 0.4750 0.4413

0.4047 0.4337 0.4446 0.4447

0.4405 0.4561 0.4716 0.4665

0.4011 0.4180 0.4686 0.5461

0.4536 0.3321 0.3720 0.2813

SiC/L40/CLD

NOTES: TC2 — Thermocouple located after the fuff slightly above the rotor. TC3 — Thermocouple located before the tuft slightly above the rotor DL — Drag Load (lbs. @ 6.085 in. radius)

 Silicon Carbide Fibers Sic

20 - 20,000 RPM's 30 - 30,000 RPM's 40 - 40,000 RPM's

(7.5 scfm) GLD - Cold Air Σ

Small Weight 164.30gI – Medium Weight 259.73gLarge Weight 493.56

 Coefficient of Friction ß

FILE: TTRPBCSI

SiC/L30/CLD

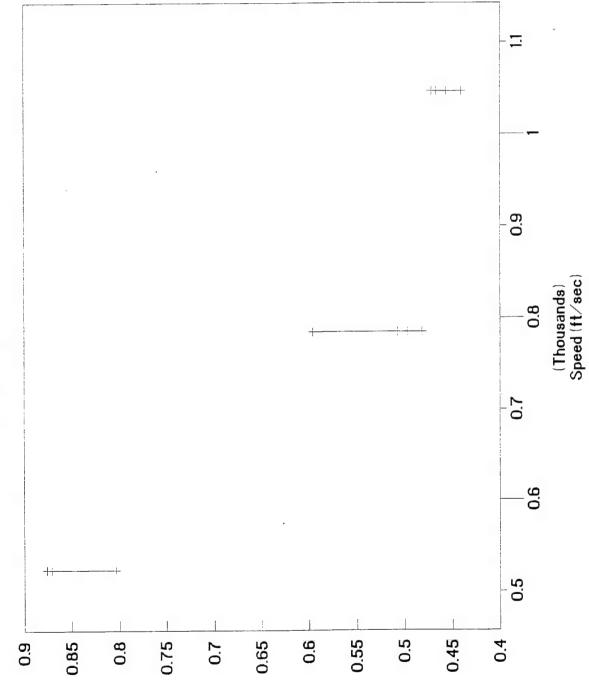
SiC/S40/CLD

SiC/M30/CLD

SiC/S30/CLD

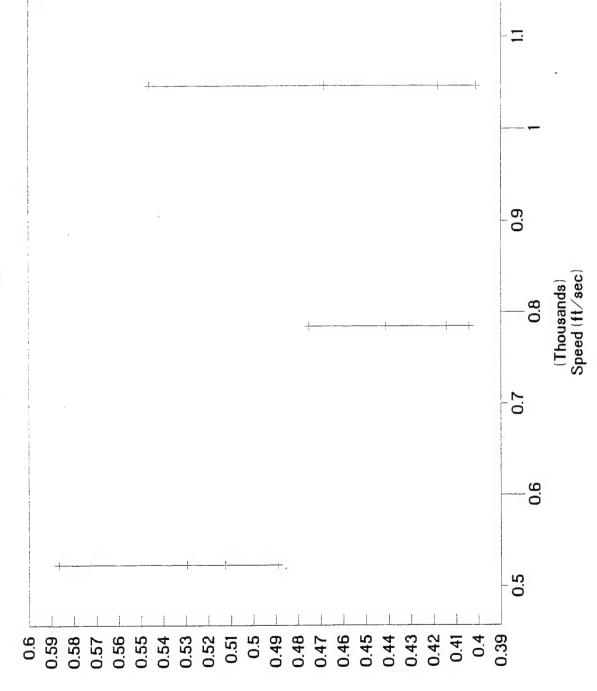
SiC/L20/CLD

SiC on BN/PSZ Cold Air(7.5 scfm) / Small Wt.



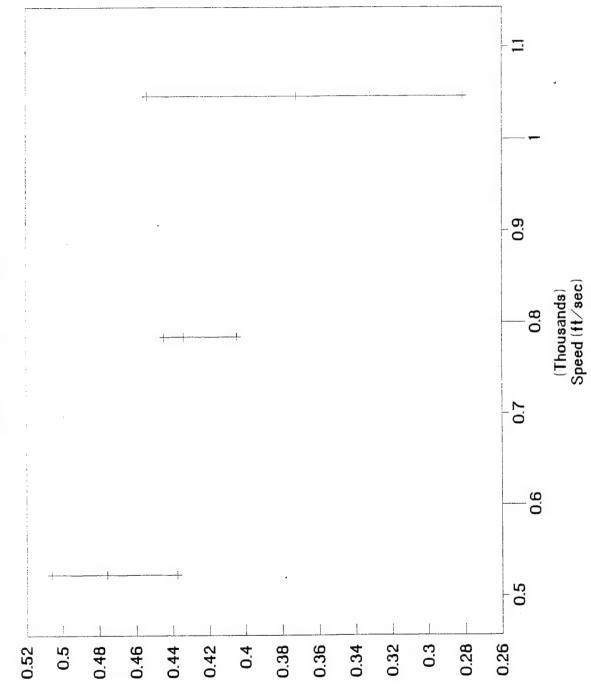
CofF

SiC on BN/PSZ Cold Air(7.5 scfm) / Medium Wt.



CofF

SiC on BN/PSZ Cold Air(7.5 acfm) / Large Wt.



CofF

PART #8 TRAC A PN 036 20000 HPMS PART #8 TRAC B PN 037 30000 HPMS PART #8 TRAC C PN 037 40000 RPMS

PSZ COATED ROTOR

AMBIENT AIR TESTING

| START TEMP END START TC2(°F) TC2(°F) TC3(°F) T | 2 | TEMP CHANGE (P.) 15.1 | S | START DL C 0.007 | CHANGE TE | START IT WT(g) T | END TFT WT(g) 13.3510 | 0.0036 0.0076 | (lb) 0.2494 | (lb) 0.3917 | Cf 0.6368 |
|--|------|-----------------------------|-----------|------------------------|----------------|---------------------|-----------------------------|------------------|----------------|----------------|--------------|
| 79.7 68.2 | - | 11.5 | 0.126 | 0.003 | 0.123 | 13.3480 | 13.3456 | 0.0024 | 0.2494 | 0.3917 | 0.6368 |
| 73.1 17.9 80.2 68.0 | 12 | 7 | | 0.003 | 0.118 | 13.3450 | 13.3430 | 0.0020 | 0.2393 | 0.3917 | 0.010 |
| 104.5 63.8 40.7 92.9 00.0 24.1 104.9 99.3 72.2 20.1 | 20. | - | | 0.005 | 0.174 | 13.3420 | 13.3384 | 0.0036 | 0.3529 | 0.6021 | 0.5861 |
| 82.0 23.6 92.0 73.7 | 18 | 18.3 | 0.163 | 900.0 | 0.157 | 13.3384 | 13.3345 | 0.0039 | 0.3184 | 0.6021 | 0.5288 |
| 83.2 21.4 90.9 74.2 | | 16.7 | | 0.004 | 0.161 | 13.3345 | 13.3321 | 0.0024 | 0.3265 | 0.6021 | 0.5423 |
| 62.6 116.6 71.2 | | 45.4 | 0.323 | 900.0 | 0.317 | 13.3321 | 13.2630 | 0.0691 | 0.6429 | 1.1175 | 0.5753 |
| 94.2 34.9 116.1 75.9 | 4 | 40.2 | | 9000 | 0.275 | 13.2630 | 13.2312 | 0.0318 | 0.5577 | 1.1175 | 0.4991 |
| 116.1 94.0 22.1 103.0 77.3 28 | N C | 25.7 | 0.245 | 0.001 | 0.244 0.258 | 13.2312 | 13.2012 | 0.0300 | 0.4948 | 1.11/4 | 0.4428 |
| 93.0 | 3 | | | | | | | | | | |
| 89.9 29.3 108.0 95.8 | 12 | NIC | | 0.007 | 0.039 | 13.0622 | 13.0606 | 0.0016 | 0.0793 | 0.3911 | 0.2022 |
| 101.2 24.5 113.5 94.2 | 19.3 | | | 0.000 | 0.093 | 3.0000 | 13.0390 | 0.0016 | 0.1880 | 0.391 | 0.4823 |
| 132.2 102.3 29.9 116.0 94.7 21.3 | 21.3 | | | 0.003 | 0.137 | 13.0590 | 13.0584 | 00000 | 0.2778 | 0.3911 | 200 |
| 81.3 64.0 121.3 94.9 | 26.4 | | 0.185 | 0.008 | 0.177 | 13.0584 | 13.0515 | 0.0009 | 0.3550 | 0.3911 | 0.3179 |
| 96.5 | 27.3 | | | 0.007 | 0.033 | 13.0515 | 13.0438 | 0.0077 | 0.0669 | 0.6015 | 0.1113 |
| 113.6 26.1 119.8 98.1 | 21.7 | | | 600.0 | 0.023 | 13.0143 | 13.0099 | 0.0044 | 0.0466 | 0.6014 | 0.0776 |
| 128.7 110.2 18.5 111.8 98.6 13.2 | 13.2 | | 0.027 | 0.010 | 0.017 | 13.0099 | 13.0047 | 0.0052 | 0.0345 | 0.6014 | 0.0573 |
| 109.0 13.2 108.1 98.5 | 9.6 | | i | 600.0 | 990.0 | 13.0047 | 13.0000 | 0.0047 | 0.1338 | 0.6014 | 0.2226 |
| 79.4 49.4 124.3 | 31.0 | | 0.206 | 600.0 | 0.197 | 13.0000 | 12.9608 | 0.0392 | 0.3995 | 1.1169 | 0.3577 |
| . 122.8 99.2 | 23 | 9 | 0.081 | 0.008 | 0.073 | 12.9608 | 12.9146 | 0.0462 | 0.1480 | 1.1168 | 0.1326 |
| 117.9 27.1 128.5 100.1 | 28. | * | 0.094 | 0.009 | 0.085 | 12.9146 | 12.8510 | 0.0636 | 0.1724 | 1.1166 | 0.1544 |
| 43.1 | 30. | 8 | 0.225 | 0.009 | 0.216 | 13.2494 | 13.1849 | 0.0645 | 0.4380 | 1.1174 | 0.3920 |
| 106.0 | 12.6 | - | 0.104 | 0.007 | 260.0 | 13.1819 | 13.1760 | 0.0059 | 0.1967 | 0.3913 | 0.5027 |
| | 8 | 2 | 0.102 | 0.008 | 0.094 | 13.1760 | 13.1755 | 0.0005 | 0.1906 | 0.3913 | 0.4871 |
| 116.5 20.1 117.7 110.4 | 7.3 | 9 | 0.108 | 0.008 | 0.100 | 13.1755 | 13.1720 | 0.0035 | 0.2028 | 0.3913 | 0.5182 |
| 20.0 | 10.8 | | 0.110 | 0.007 | 0.103 | 13.1720 | 13.1680 | 0.0040 | 0.2089 | 0.3913 | 0.5338 |
| 98.4 39.9 121.4 | 11.1 | _ | 0.113 | 0.008 | 0.105 | 13.1680 | 13.1596 | | 0.2129 | 0.6017 | 0.3539 |
| 119 4 211 1221 1130 | 9.1 | į | 0.137 | 0.008 | 0.129 | 13.1596 | 13.1465 | | 0.2616 | 0.6017 | 0.4348 |
| 1202 | 8.7 | Ļ | 0.146 | 0.007 | 0.139 | 13.1465 | 13.1207 | | 0.2819 | 0.6016 | 0.4685 |
| 136.6 119.4 17.2 119.6 111.1 8. | 8 | 2 | 0.148 | 900.0 | 0.142 | 13.1207 | 13.0946 | 0.0261 | 0.2880 | 0.6016 | 0.4787 |
| | | | | | | | | | | | |
| DISC DISC | SC | ā | DISC DISC | | DISC | | DISC | 0.0000 | 0.0000 | 1.0883 | 00000 |
| DISC DISC DISC | သွ | 5 | | | | | SC | 0.0000 | 0.0000 | 1.0863 | 0.0000 |
| DISC | 2 | 5 (| חמכו | | | | 200 | 0.000 | 3000 | .0003 | 0.000 |
| DISC DISC DISC | 2 | 2 | | | | S S S | 3 | 2000 | 333 | 2002 | O COCC |

NOTES: TC2 — Thermocouple located after the tuft slightly above the rotor. TC3 — Thermocouple located before the tuft slightly above the rotor DL — Drag Load (lbs. @ 6.085 in. radius)

SiC - Silicon Carbide Fibers

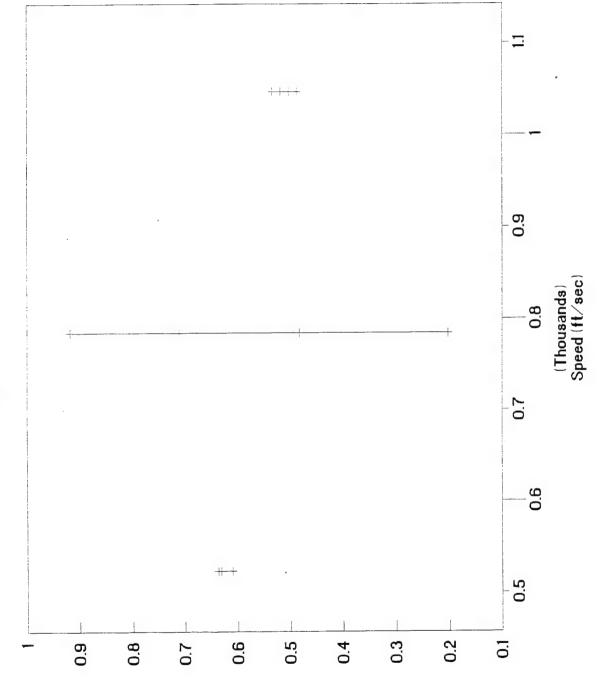
20 - 20,000 RPM's 30 - 30,000 RPM's 40 - 40,000 RPM's

AMB - Ambient Air (no flow)

S – Small Weight 164.30g M – Medium Weight 259.73g L – Large Weight 493.56

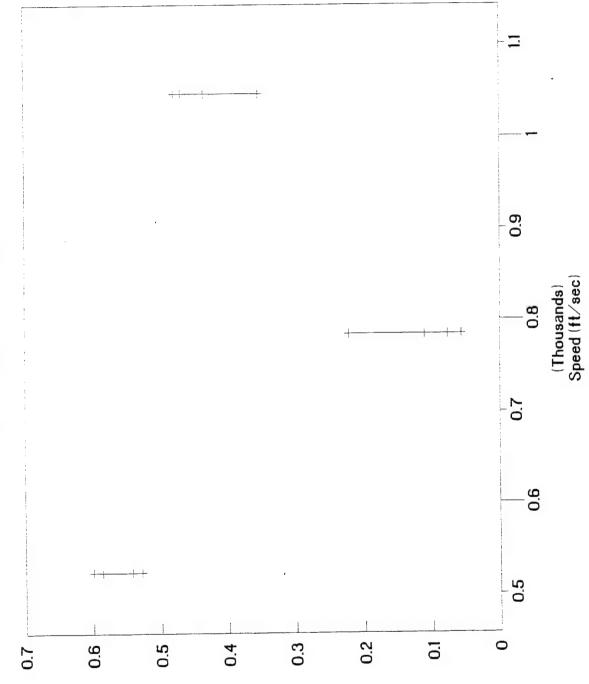
CF - Coefficient of Friction

SiC on PSZ Ambient Air / Small Wt.



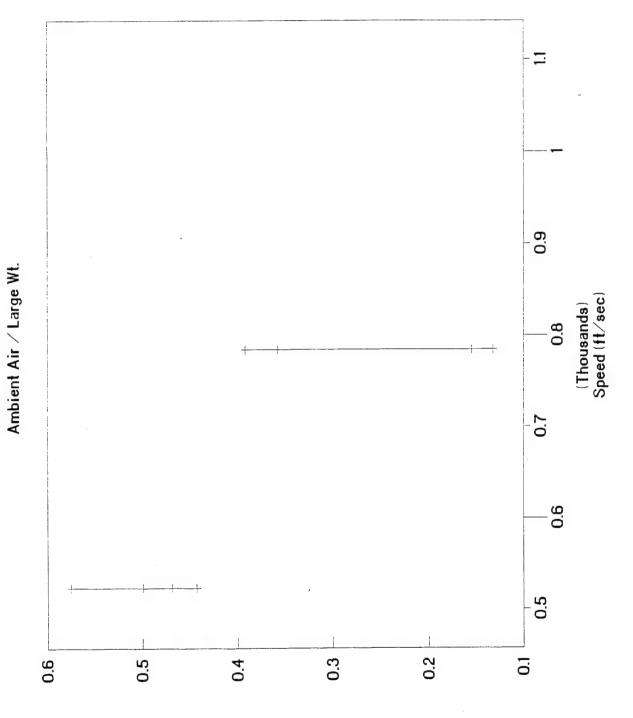
CofF

SiC on PSZ Ambient Air / Medium Wt.



CofF

SiC on PSZ Ambient Air / Large Wt.



CofF

CHROME CARBIDE COATED ROTOR

AMBIENT AIR TESTING

PN 019; SIC/S20 THRU SIC/L30 PN 020 REMAINDER

PART #7 TRACK A

| 1 | | | | | - | * | | | | | | | | | | 1 | | - 1 | | -1- | | | 1= | _ | | | 7. | 1== | | | | | 10 | 1-1 | T. | - i | -1 | 211 | |
|------------------|-------------|---------|---------|---------|---|-------------|---------|---------|---------|-------------|---------|---------|---------|-------------|--------------|---------|---------|---------|-------------|--------|---------|-----|-------------|---------|---------|---------|-------------|---------|---------|---------|-------------|---------------|---------|---------|--------|-------------|---------|----------|---------|
| ਹ | 0.3732 | 0.2851 | 0.2592 | 0.2384 | | 1.1325 | 0.7382 | 0.7820 | 0.9741 | 0.4520 | 0.4484 | 0.5464 | 0.5809 | 0.4044 | 0.4407 | 0.3266 | 0.3370 | 70000 | 866 | 0.300 | 0.2032 | | 0.3050 | 0.2960 | 0.2143 | 0.3668 | 0.3167 | 0.3063 | 0.2856 | 0.2959 | 0.2902 | 0.2767 | 0.283 | 0.2700 | | 0.2688 | 0.3324 | 0.325 | 0.327 |
| N (dl) | 0.3913 | 0.3913 | 0.3912 | 0.3912 | | 0.6017 | 0.6017 | 0.6017 | 0.6017 | 1.1172 | 1.1172 | 1.1172 | 1,1172 | 0.3912 | 0.3911 | 0.3911 | 0.3911 | 2000 | 2 4 | 2000 | 0.00 | | 1.1170 | 1.1169 | 1.1167 | 1.1167 | 0.3906 | 0.3906 | 0.3906 | 0.3906 | 0.6010 | 0.6010 | 0.6009 | 0.6009 | 1011 | 50 | 1.1164 | 1.153 | 1.1162 |
| DRAG (Ib) | 0.1460 | 0.1115 | 0.1014 | 0.0933 | | 0.6814 | 0.4441 | 0.4705 | 0.5861 | 0.5050 | 0.5009 | 0.6104 | 0.6490 | 0.1582 | 0.1724 | 0.1278 | 0.1318 | 3000 | 2000 | 0.1000 | 1480 | | 0.3407 | 9088.0 | 0.2393 | 0.4097 | 0.1237 | 0.1197 | 0.1115 | 0.1156 | 0.1744 | 0.1663 | 0.1704 | 0.1622 | 40000 | 0.3250 | 0.3711 | 0.3630 | 0.3650 |
| WT OSS(9) | 0.0018 | 0.0006 | 9000.0 | 0.0000 | | 0.0015 | 9000.0 | 0.0002 | 0.0012 | 0.0130 | 0.0038 | 0.0080 | 0.0056 | 0.0052 | 0.0036 | 0.0040 | 9:00:0 | 22.00 0 | 38 | 0000 | 0.000 | | 0.0265 | 0.0325 | 0.0397 | 0.0247 | 0.0116 | 0.0074 | 0.0084 | 0.0070 | 0.0137 | 0.0111 | 0.0121 | 0.0130 | 0 00 0 | 0.0346 | 0.0337 | 0.0333 | 0.0369 |
| END FT WT(9) | 13,1400 | 13,1395 | 13.1336 | 13.1336 | | 13.1321 | 13.1316 | 13,1314 | 13.1302 | 13.1172 | 13,1134 | 13.1054 | 13.0398 | 13.0946 | 13.0910 | 13.0864 | 13.0828 | 1200260 | 30000 | 3.0000 | 13.0370 | | 13.0205 | 12.9880 | 12.9007 | 12.8760 | 12.8642 | 12.8568 | 12.8484 | 12.8414 | 12,8277 | 12.8166 | 12.8045 | 12.7915 | 00000 | 12.7363 | 12.7232 | 12.68/9 | 12,6510 |
| START | 13.1418 | 13.1400 | 13.1342 | 13.1336 | | 13.1336 | 13.1321 | 13,1316 | 13.1314 | 13.1302 | 13.1172 | 13.1134 | 13.1064 | 13.0998 | 13.0946 | 13.0904 | 13.0864 | 42 0000 | 13.0020 | 3.070 | 13.0000 | | 13.0470 | 13.0205 | 12.9404 | 12.9007 | 12.8758 | 12.8642 | 12.8568 | 12.8484 | 12.8414 | 12.8277 | 12.8166 | 12.8045 | 70.70 | 12.7315 | 12.7569 | 12.7.232 | 12.6879 |
| DL CHANGE TI | 0.072 | 0.055 | 0.060 | 0.046 | | 0.336 | 0.219 | 0.232 | 0.289 | 0.249 | 0.247 | 0.301 | 0.320 | 0.078 | 1 | 0.063 | 1 1 | 000 | 000 | 0000 | 0.004 | | 0.168 | 0.163 | 0.118 | 0.202 | 0.061 | 0.069 | 0.055 | 0.057 | 0.086 | 0.082 | 0.084 | 0.080 | 0.10 | 0.159 | 0.183 | 0.179 | 0.180 |
| START DL (| ш | 0.003 | 0.005 | 900.0 | | 0.004 | 900.0 | 0.003 | 0.004 | 0.004 | 0.00 | 900.0 | 0.007 | 0.007 | 0.007 | 0.008 | 0.005 | 000 | 300 | 880 | 0.00 | | 0.003 | 0.005 | 0.005 | 0.004 | 0.003 | 0.002 | 0.003 | 0.003 | 0.004 | 0.004 | 0.003 | 0.004 | 000 | 0.002 | 0.003 | 0.003 | 0.003 |
| END DI. | 0.076 | 0.058 | 0.055 | 0.062 | | 0.340 | 0.225 | 0.235 | 0.293 | 0.253 | 0.351 | 0.307 | 0.327 | 0.085 | 0.092 | 0.071 | 0.070 | 900 | 38 | 100 | 0.003 | | 0.171 | 0.168 | 0.123 | 907.0 | 0.064 | 0.061 | 0.058 | 0.060 | 060.0 | 0.086 | 0.087 | 0.084 | 2 | 0.161 | 0.186 | 281.0 | 0.183 |
| TEMP ANGE(*F) | 10.8 | 7.1 | 6.7 | 4.8 | | 50.2 | 23.4 | 28.1 | 36.4 | 40.2 | 35.7 | 43.1 | 39.0 | 11.8 | 6.6 | 6.9 | 9.5 | 17.4 | 1.00 | 10.0 | 0.0 | | 25.3 | 25.8 | 39.9 | 22.6 | 3.0 | 7.2 | 6.5 | 1.9 | 1.8 | 7.6 | 3.8 | 9.5 | | 17.0 | 0.9 | 13.7 | 14.3 |
| START TEMP | 79.3 | 80.0 | 76.0 | 76.2 | | 74.7 | 78.4 | 75.0 | 74.0 | 74.9 | 76.0 | 77.8 | 79.3 | 96.5 | 99.0 | 99.11 | 98.7 | 200 | 93.5 | 33.0 | 2.00 | 3 | 99.4 | 98.8 | 100.9 | 108.0 | 126.2 | 122.8 | 124.5 | 125.4 | 124 1 | 124.4 | 127.2 | 128.3 | | 131.0 | 137.5 | 131.0 | 130.2 |
| END TC3(*F) | 90.1 | 97.1 | 82.7 | 81.0 | | 124.9 | 101.8 | 103.1 | 110.4 | 115.1 | 111.7 | 120.9 | 118.3 | 108.3 | 108.9 | 106.0 | 106.9 | 0 077 | 0.0 | 9.0 | 100.0 | 2 | . 124.7 | 124.6 | 140.8 | 130.6 | 129.2 | 130.0 | 131.0 | 127.3 | 139.9 | 132.0 | 131.0 | 137.8 | | 148.0 | 145.5 | 146.7 | 144.5 |
| TEMP NGE("F) | 19.0 | 8.8 | 12.2 | 6.7 | | 55.6 | 15.6 | 25.3 | 38.4 | 53.2 | 41.6 | 60.0 | 41.6 | 23.6 | 13.9 | 4.6 | 14.3 | 200 | 34.0 | 0.22 | 9.61 | 1 | 33.7 | 41.7 | 59.5 | 35.2 | 20.6 | 13.6 | 14.4 | 10.4 | 17.7 | 15.1 | 12.2 | 26.4 | | 27.5 | 21.7 | 24.5 | 22.6 |
| START TEMP | 71.5 | 80.0 | 70.4 | 74.8 | | 75.2 | 91.0 | 82.1 | 78.0 | 81.8 | 85.0 | 71.0 | 88.3 | 94.2 | 103.5 | 103.4 | 99.0 | | 04.0 | 99.0 | 2.5 | 3 | 99.5 | 91.1 | 103.2 | 125.2 | 117.6 | 123.1 | 124.0 | 124.7 | 1936 | 2000 | 127.3 | 117.9 | | 131.1 | 134.0 | 133.9 | 134.7 |
| END TC2(*F) | L | 88.8 | 82.6 | 81.5 | | 130.8 | 106.6 | 107.4 | 116.4 | 135.0 | 126.6 | 131.0 | 129.9 | 117.8 | 117.4 | 1128 | 113.3 | | 118.5 | 121.0 | 119.3 | 0.1 | 133.2 | 132.8 | 162.7 | 160.4 | 138.2 | 136.7 | 138.4 | 136.1 | 1413 | 140 6 | 139.5 | 144.3 | | 158.6 | 155.7 | 158.4 | 157.3 |
| | SIC/S20/AMB | | | | | SIC/M20/AMB | | | | SIC/L20/AMB | | | | SICISANIAMB | and and fair | | | | SIC/M30/AMB | | | | SIC/L30/AMB | | | | SIC/S40/AMB | | | | SIC/MAD/ANB | Cimilar Maria | | | | SiC/L40/AMB | | | |

NOTES: TC2 – Thermocouple located after the tuft slightly above the rotor. TC3 – Thermocouple located before the tuft slightly above the rotor DL – Drag Load (Ibs @ 6.095 in.radius)

SIC - Silicon Carbide Fibers

20 - 20,000 RPM's 30 - 30,000 RPM's 40 - 40,000 RPM's

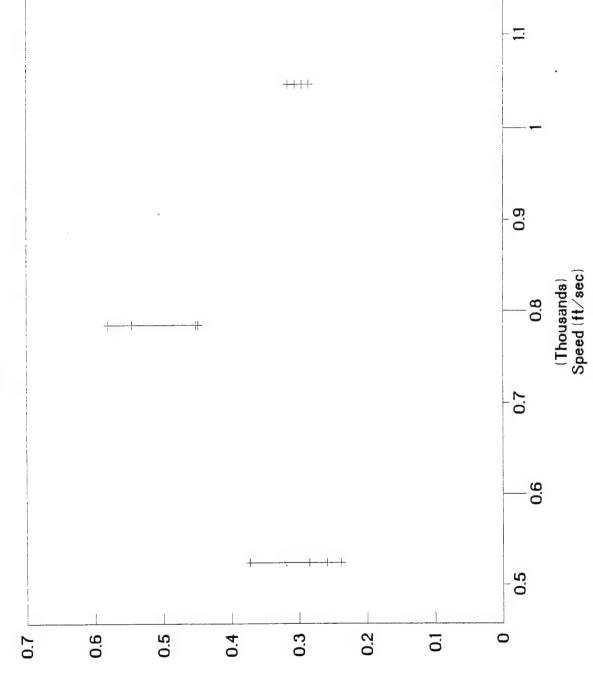
AMB - Ambient Air (no flow)

*** - Test Failure

S – Small Weight 164.30g M – Medlum Weight 259.73g L – Large Weight 493.56

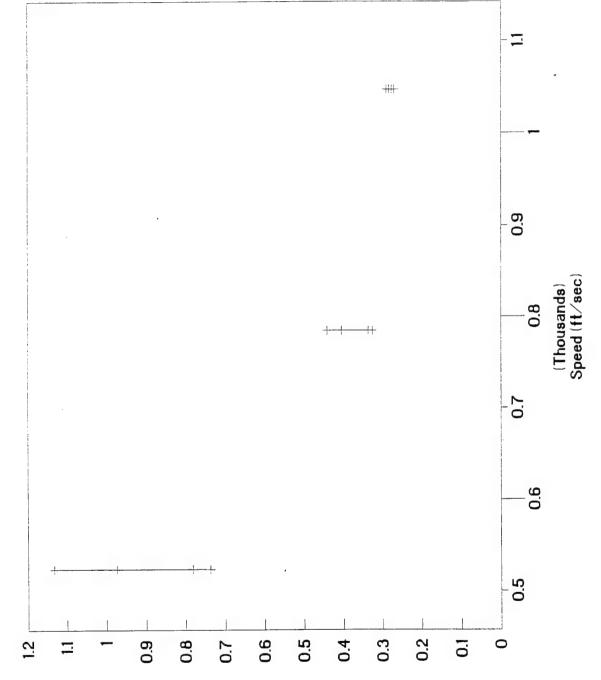
CF - Coefficient of Friotion

SiC on Chrome Carbide



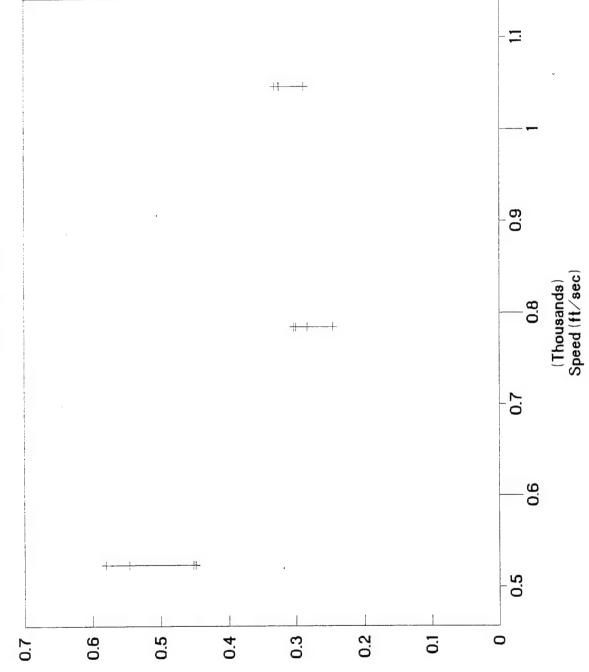
C of F

SiC on Chrome Carbide



CofF

SiC on Chrome Carbide Ambient Air / Large Wt.



CHROME CARBIDE COATED ROTOR

PN 025 SIC/S20 - SIC/S40(1) PN 035 LAST THREE

PART #7 THACK B

COLD AIR TESTING

0.2647 0.3477 0.3373 0.2751 0.4150 0.5533 0.5499 0.5972 0.3959 0.4014 0.4559 0.4541 0.3741 0.3845 0.3741 0.3533 0.3815 0.3748 0.3917 0.3816 0.3307 0.2980 0.2798 0.2472 0.4179 0.4052 0.4270 0.4179 0.2867 0.2159 0.3104 0.2969 0.4256 0.6696 0.6281 \overline{c} 811.1 811.1 1.1163 0.3904 0.6007 0.6012 0.6012 0.6012 0.6012 1.1162 1.1162 1.1161 1.1161 0.3907 0.3907 0.3907 0.6011 0.6011 0.6011 1.1166 1.1165 1.1165 1.1164 0.2292 0.2251 0.2352 0.2292 0.1460 0.1501 0.1460 0.1034 0.1359 0.1318 0.1075 0.2494 0.3326 0.3306 0.3590 0.3691 0.3326 0.3123 0.2758 0.1724 0.1298 0.1866 0.1785 0.4664 0.4522 0.4766 0.4664 0.1663 0.2616 0.2454 0.1420 0.4421 0.5030 0.5070 0.0128 0.0126 0.0136 0.0456 0.0456 0.0509 0.0851 0.0110 0.0084 0.0128 0.0145 0.0034 0.0021 0.0021 0.00010 0.0000 0.0024 0.0094 0.0056 0.0013 0.0047 0.0350 0.0246 0.0305 0.0292 0.0009 ₹ CHANGE IFT WT(g) IFT WT(g) 1 0.0651 12.9158 12.9137 0.067 12.9137 12.9127 0.065 12.9127 12.9120 2 12.8907 12.8883 12.8883 12.8883 1 12.8883 12.8874 1 12.8874 12.8667 12.7340 12.7258 12.7164 12.7074 12.5606 12.6767 12.6258 12.5407 12.9120 12.9111 12.9107 12.9096 12.9096 12.9096 12.6525 12.6415 12.6415 12.6331 12.6331 12.6203 12.6203 12.6508 12.8773 12.8717 12.8704 12.8657 12.8307 12.8061 12.7756 12.7464
 0.113
 12.7074
 12.6946

 0.111
 12.6946
 12.6820

 0.116
 12.6820
 12.6684

 0.113
 12.6684
 12.6525
 12.867 12.8773 12.8717 12.8704 12.6058 12.7223 12.6767 12.6258 12.8657 12.8307 12.8061 12.7756 12.7464 12.7340 12.7258 0, 123 0, 164 0, 163 0, 177 0.085 0.230 0.082 0.129 0.121 0.070 0.218 0.221 0.251 0.250 0.072 0.072 0.068 0. 182 0. 154 0. 136 \$1ART DL 0.004 0.005 0.006 0.006 0.008 0.007 0.007 0.007 0.007 0.006 0.007 0.007 0.007 0.007 0.007 0.006 0.006 0.225 0.228 0.258 0.257 0.076 0.121 0.118 0.123 0.120 0. 190 0. 172 0. 163 0. 146 END DL 0.055 0.071 0.057 0.089 0.237 0.089 0.135 0.076 0. 130 0. 171 0. 169 0. 184 TC3CF) CHANGE(F) 73.0 0.1 73.1 0.1 73.8 0.4 74.0 0.0 0 0 0 0 4 0.0 0.00 -0.2 0 0 0 0 0.4 0.8 1.6 0.7 0.4 TEMP 65.6 65.9 66.5 66.4 68.9 70.7 70.7 71.8 63.6 65.3 66.2 65.9 73.9 61.2 61.3 61.8 60.9 60.2 60.1 60.5 59.5 60.3 60.2 60.0 58.9 60.0 60.4 59.7 65.4 65.9 66.3 60.1 59.5 59.5 60.3 60.9 60.6 60.5 60.5 60.7 60.8 60.4 65.8 65.8 65.8 70.2 70.8 70.8 71.4 61.4 61.7 62.0 73.9 73.2 74.2 74.0 74.2 77.3 77.3 75.1 8.1 78.8 4.1 31.7 19.6 19.7 14.1 22.2 10.9 11.6 10.5 12.8 6.2 6.6 22.8 24.6 27.1 22.9 39.9 30.2 33.4 22.7 27.3 13.0 11.2 12.1 6.1 94.0 94.1 95.4 95.4 78.6 95.8 102.2 106.1 74.2 91.3 87.6 91.3 78.8 80.0 75.8 81.0 61.7 73.0 75.0 76.3 74.0 76.2 80.2 76.3 73.7 85.4 85.1 87.6 76.9 92.2 94.0 93.6 104.2 105.2 105.2 105.2 108.1 108.6 110.1 123.2 127.5 121.8 125.8 95.6 95.6 92.9 96.5 110.0 112.2 110.5 121.5 121.0 121.0 62(°F) 81.3 82.7 83.2 82.9 94.9 96.8 96.7 83.9 83.9 86.6 86.8 SIC/M40/CLD TEST SIC/S20/CLD SIC/M30/CLD SIC/M20/CLD SIC/S30/CLD SIC/S40/CLD SIC/L40/CLD SIC/L 20/CLD SIC/L30/CLD FIBER/

NOTES:
TC2 - Thermocouple located after the tuft slightly above the rotor.
TC3 - Thermocouple located before the tuft slightly above the rotor IC3 - Thermocouple located before in tall slightly above the rotor II - Drag Load (lbs @ 6.085 in radius)

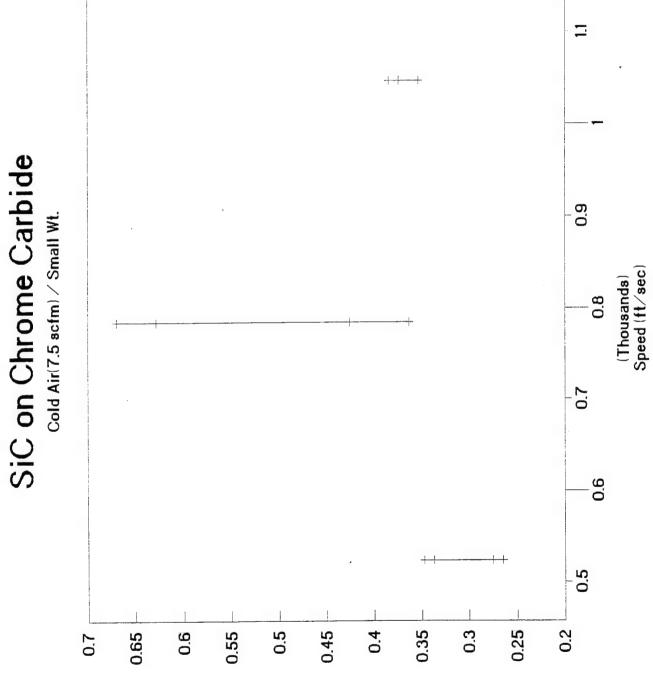
Fibers - Silicon Carbide Sic

(7.5 SCFM) - COLD AIR CLD တ Σ

Small Weight 164.30g
 Medium Weight 259.73g
 Large Weight 493.56

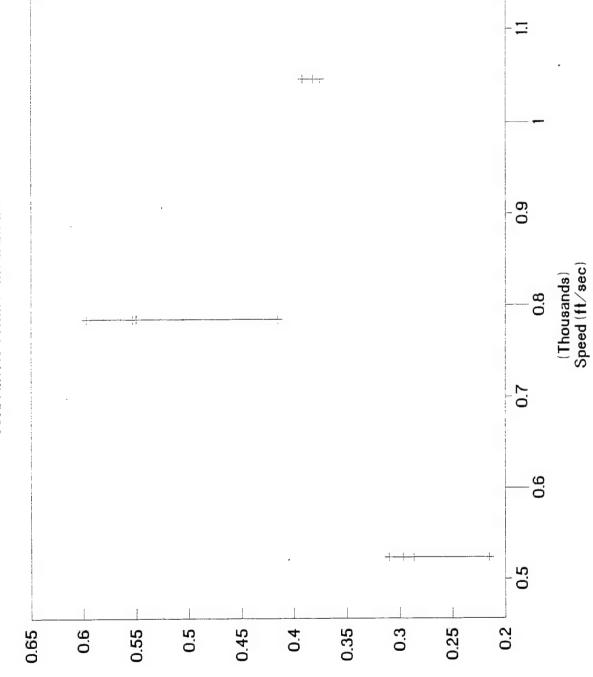
Coefficient of Friction ł Ç,

FILE: TTRPCCSI



CotF

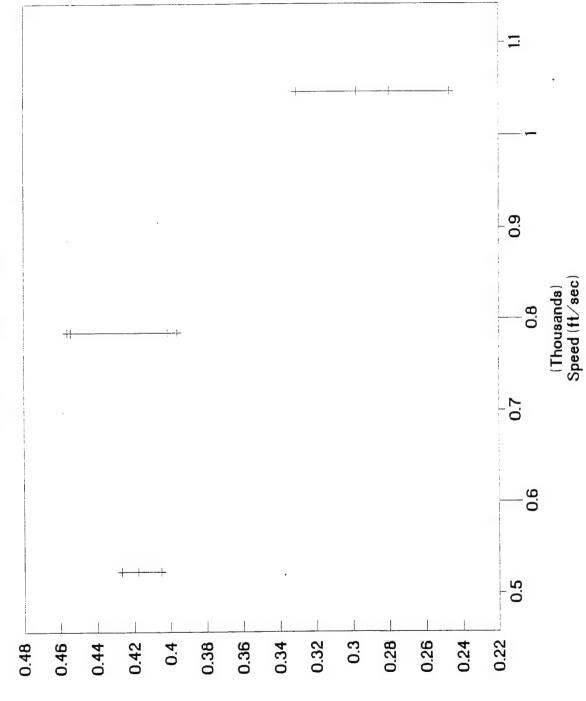
SiC on Chrome Carbide Cold Air(7.5 scfm) / Medium Wt.



CotF

SiC on Chrome Carbide Cold Air(7.5 scfm) / Large Wt.





CofF

PART # 10A TUFT # 050--START THRU M40(1) TUFT # 051--REMAINDER

BARE ROTOR

AMBIANT AIR TESTING

 $\begin{tabular}{ll} NOTES: & TC2 - Thermocouple located after the tuft slightly above the rotor. TC3 - Thermocouple located before the tuft slightly above the rotor DL - Drag Load (lbs. @ 6.085 in. radius) \\ \end{tabular}$

SiC - Silicon Carbide Fibers

20 - 20,000 RPM's 30 - 30,000 RPM's 40 - 40,000 RPM's

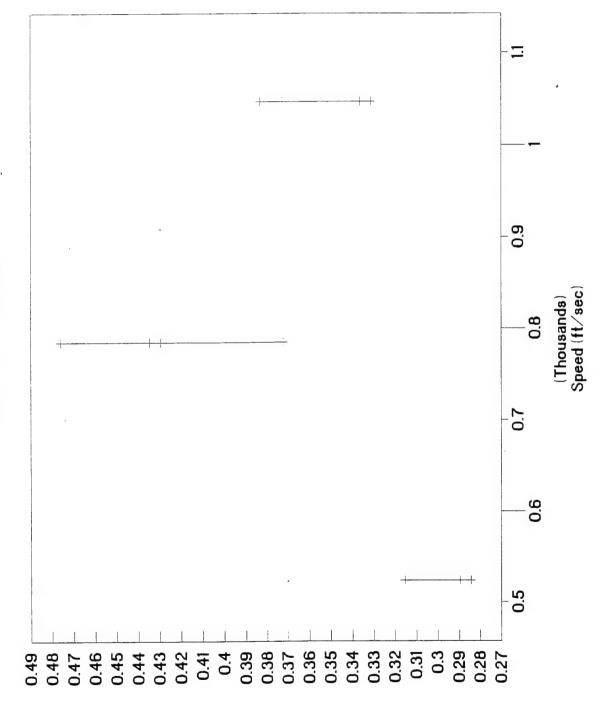
AMB -- Ambiant Air (no flow)

S – Small Weight 164.30g M – Medium Weight 259.73g L – Large Weight 493.56

CF - Coefficient of Friction

FILE: TTRPNASI

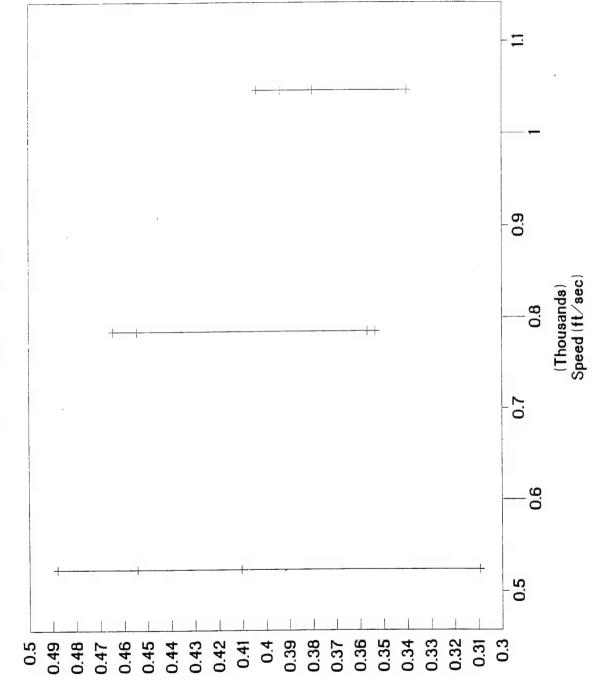
SiC on Bare Rotor



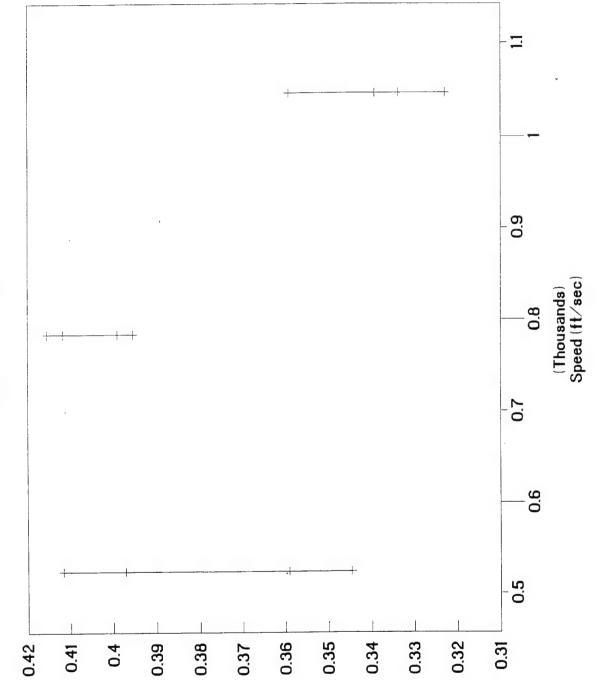
CofF

SiC on Bare Rotor

Ambient Air / Medium Wt.



SiC on Bare Rotor



C of F

BARE ROTOR

| | | ų, | | | | | BARE | BARE ROTOR | | | | | | | |
|---|---|-------------------|--------|---------|----------|-------------------|-------------|---------------------|----------|----------|---------|---------|--------|---------|--------|
| TUFT #052 - Start thru Tuft #053 - Remainder | Start thru MD/40 Remainder | 0/40 | | | | ∢ | MBIANT A | AMBIANT AIR TESTING | , | | | | | | |
| FIBER/ | END | START | TEMP | END | START | TEMP | END | _ | DL | START | | W | DRAG | z | Cf |
| TEST | ICZ(F) | TC2(*F)CHANGE(*F) | ANGECE | IC3(:F) | TC3(*F)C | C3(*E) CHANGE(*E) | Id | 1- | \vdash | FT WT(g) | - 1 | 1088(9) | (41) | (115) | 0 0740 |
| H25/S20/AMB | 94 1 | 88.9 | 5.2 | 4 101 | 93.0 | 0 0 | 0 054 | 0.003 | 0.053 | 13.4308 | 13.4204 | 0.0104 | 0.1034 | 0.3919 | 0.2639 |
| | 94.0 | 92.4 | 13 | 101 | 95.5 | 5.5 | 0.049 | 0.002 | 1 | 13.4204 | | 0.0053 | 0.0953 | 0.3919 | 0.2432 |
| | 93.7 | 92.0 | 1.7 | 1.01.7 | 96.0 | . 5.7 | 0.045 | 0.001 | 0.044 | 13.4151 | 13.4120 | 0.0031 | 0.0892 | 0.3919 | 0.2277 |
| H25/M20/AMB | 95.3 | 92.0 | 3.3 | 105.0 | 96.4 | 8.6 | 0.062 | 0.001 | 0.061 | 13.4120 | 13.4078 | 0.0042 | 0.1237 | 0.6023 | 0.2054 |
| Carried Control | 95.3 | 93.4 | 1.9 | 104.2 | 0.86 | 6.2 | 0.063 | 0.000 | 1 | 13.4078 | ΙI | 0.0030 | 0.1278 | 0.6023 | 0.2121 |
| | 96.2 | 93.4 | 2.8 | 104.9 | 7.96 | 8.2 | 0.060 | 0.002 | | 13.4048 | | 0.0026 | 0.1176 | 0.6023 | 0.1953 |
| | 96.4 | 94.2 | 2.2 | 105.2 | 98.3 | 6.9 | 0.062 | 0.002 | 0.060 | 13.4022 | 13.4006 | 0.0016 | 0.1217 | 0.6023 | 0.2020 |
| H25/120/AMB | 100.3 | 94.3 | 0.9 | 114.7 | 97.4 | 17.3 | 0.126 | 0.002 | | 13.4006 | 1 1 | 0.0059 | 0.2515 | 1.1178 | 0.2250 |
| | 103.1 | 99.5 | 3.6 | 117.0 | 100.4 | 16.6 | 0.119 | 0.002 | | 13.3947 | 13,3915 | 0.0035 | 0.2373 | 1.1178 | 0.2123 |
| | 102.8 | 101.5 | 6. | 119.8 | 102.7 | 17.1 | 0.142 | 0.002 | 0.140 | 13.3915 | | 0.0025 | 0.2839 | 1.1178 | 0.2540 |
| | 103.0 | 101.3 | 1.7 | 120.0 | 102.01 | 18.0 | 0.139 | 0.001 | _ | 13.3830 | | 0.00.0 | 0.6133 | 0/11/1 | 0.6304 |
| H25/S30/AMB | 112.5 | 101.9 | 10.6 | 127.1 | 117.8 | 9.3 | 0.040 | 0.001 | 0.039 | 13.3872 | 13.3857 | 0.0015 | 0.0791 | 0.3918 | 0.2019 |
| | 110.7 | 90.0 | 20.7 | 128.3 | 118.7 | 9.6 | 0.051 | 0.001 | 0.050 | 13.3857 | 13.3856 | 0.0001 | 0.1014 | 0.3918 | 0.2588 |
| | 113.9 | 107.8 | 6.1 | 127.8 | 122.9 | 4.9 | 0.044 | 0.003 | 0.041 | 13.3856 | 13.3839 | 0.0017 | 0.0831 | 0.3918 | 0.2122 |
| | 115.6 | 110.4 | 5.5 | 128.8 | 122.8 | 0.9 | 0 전 - | 0.002 | 0.039 | 13,3839 | 13.3824 | 0.0015 | 0.0791 | 0.3918 | 0.2019 |
| HOS/Man/AMB | 118.0 | 111.8 | 6.2 | 131.4 | 122.5 | 8.9 | 0.059 | 0.002 | 0.057 | 13.3824 | 13.3779 | 0.0045 | 0.1156 | 0.6022 | 0.1920 |
| 2000 | 1187 | 113.5 | 5.2 | 132.0 | 123.5 | 8.5 | 0.061 | 0.003 | 0.058 | 13.3779 | 13.3751 | 0.0028 | 0.1176 | 0.6022 | 0.1953 |
| - | 118.8 | 114.8 | 4.0 | 132.2 | 124.1 | 8.1 | 0.064 | 0.003 | 0.061 | 13.3751 | 13.3730 | 0.0021 | 0.1237 | 0.6022 | 0.2054 |
| | 119.1 | 114.9 | 4.2 | 132.0 | 123.9 | 8.1 | 0.063 | 0.004 | 0.059 | 13.3730 | 13.3708 | 0.0022 | 0.1197 | 0.6022 | 0.1987 |
| | 0 3,, | 0 00 | 0.50 | 143.0 | 114.0 | 0 80 | 136 | 0000 | 0.134 | 19 3708 | 13 3664 | 0 0044 | 0 2718 | 1 11 78 | 0 2431 |
| HZ5/L30/AMB | 1191 | 114.7 | 4 4 | 144.5 | 124.4 | 20.1 | 0.145 | 0.002 | 0.143 | 13.3664 | 13.3630 | 0.0034 | 0.2900 | 1.1178 | 0.2594 |
| | 123.9 | 118.5 | 5.4 | 142.7 | 125.4 | 17.3 | | 0.001 | 0.134 | 13.3630 | 13,3591 | 0.0039 | 0.2718 | 1.1178 | 0.2431 |
| | 122.1 | 119.5 | 2.6 | 146.0 | 124.7 | 21.3 | 0.149 | 0.003 | 0.146 | 13.3591 | 13.3553 | 0.0038 | 0.2961 | 1.1177 | 0.2649 |
| H25/S40/AMB | 123.6 | 108.8 | 14.8 | 141.8 | 140.3 | 1.5 | 0.031 | 0.000 | 0.031 | 13.3513 | 13.3364 | 0.0149 | 0.0629 | 0.3917 | 0.1605 |
| | 125.7 | 117.9 | 7.8 | 140.3 | 137.5 | 2.8 | 0.032 | 0.001 | 0.031 | 13.3364 | 13.3168 | 0.0196 | 0.0629 | 0.3916 | 0.1605 |
| | 126.2 | 119.5 | 6.7 | 140.7 | 137.0 | 3.7 | 0.032 | 0.001 | 0.031 | 13.3168 | 13.3024 | 0.0144 | 0.0629 | 0.3916 | 0.1605 |
| | 127.3 | 119.7 | 7.6 | 142.8 | 136.7 | 6.1 | 0.035 | 0.001 | 0.034 | 13.3024 | 13.2920 | 5 | 0.0690 | 0.3916 | 0.1701 |
| H25/M40/AMB | 129.1 | 120.9 | 8.2 | 146.7 | 137.2 | 9.6 | | | 0.045 | 13.2920 | 13.2722 | 0.0198 | 0.0913 | 0.6020 | 0.1516 |
| | 128.6 | 122.4 | 6.2 | 143.0 | 138.4 | 4.6 | | | 0.041 | 13.2687 | 13.2561 | 0.0126 | 0.0831 | 0.6019 | 0.1381 |
| | 129.1 | 122.1 | 7.0 | 144.8 | 137.8 | 7.0 | 0.042 | | 0.041 | 13.2561 | 13.2370 | 0.0191 | 0.0831 | 0.6019 | 0.1381 |
| | 1.29.1 | 122.2 | 6.9 | 145.0 | 137.8 | 7.2 | | 0.000 | 0.046 | 13.2370 | 13.2197 | 0.0173 | 0.0933 | 0.6019 | 0.1550 |
| H25/140/AMB | 112.7 | 122.3 | 9.6- | | 1 39.4 | 23.1 | | | 0.140 | 13.3330 | | 0.0240 | 0.2839 | 1.1176 | 0.2540 |
| | 116.8 | 126.3 | -9.5 | 160.8 | 142.0 | 18.8 | | | 0.115 | 13.3090 | | 0.0131 | 0.2332 | 1.1176 | 0.2087 |
| | 1.09.1 | 82.4 | 26.7 | | 127.7 | 31.9 | | 0 | 0.104 | 13.2959 | - 1 | 0.0116 | 0.2109 | 1.1176 | 0.1887 |
| | 118.0 | 121.0 | -30 | 162.0 | 141.5 | 20.5 | 0.112 | 0.000 | 0.112 | 13.2843 | 13 2749 | 0.0034 | 0.2271 | 1.1176 | 0.2032 |

NOTES: TC2 — Thermocouple located after the tuft slightly above the rolor. TC3 — Thermocouple located before the tuft slightly above the rolor DL — Drag Load (1bs @ 6.085 in.radius)

H25 - HAYNES 25 FIBERS

20 - 20,000 RPM's 30 - 30,000 RPM's 40 - 40,000 RPM's

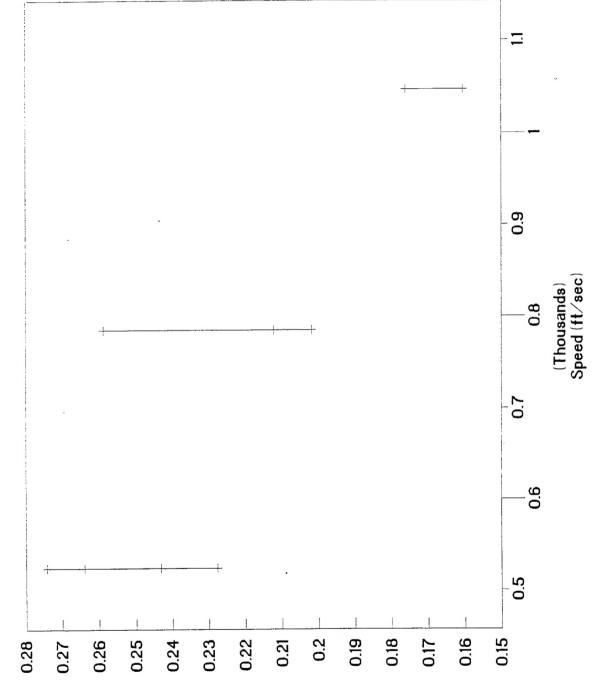
FILE: TTRPNAHA

AMB - Ambiant Air (no flow)

S – Small Weight 164.30g M – Medium Weight 259.73g L – Large Weight 493.56

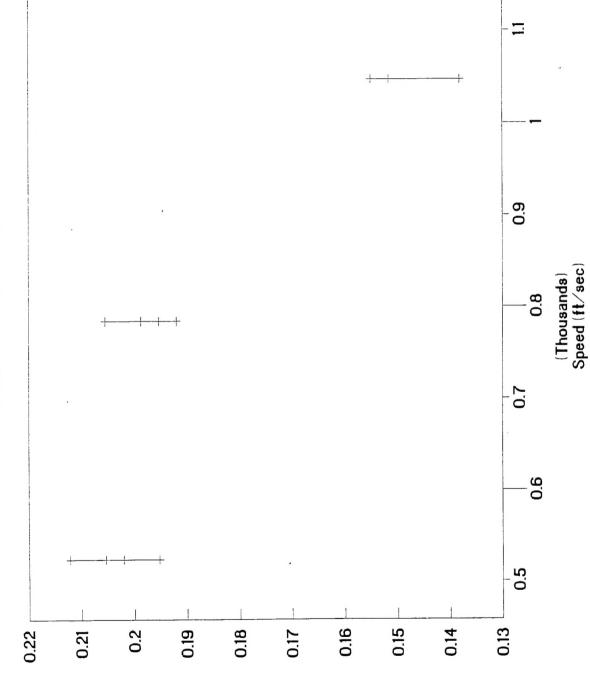
CF - Coefficient of Friction

Haynes 25 On Bare Rotor



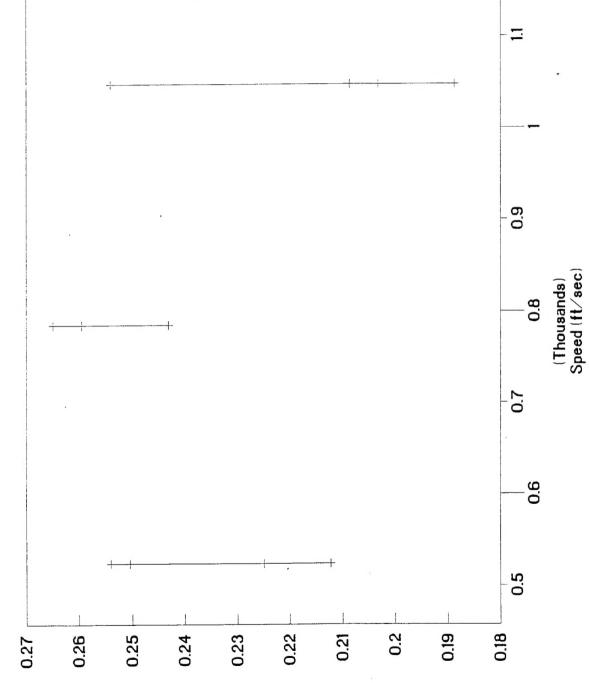
CofF

Haynes 25 On Bare Rotor



C of F

Haynes 25 On Bare Rotor Ambient Air / Large Wt.



CofF